

June, 2004

ICS 420-1

PRIMARY MOBILE SUPPRESSION RESOURCES (continued)

RESOURCE	RADIO CALL	COMPONENTS	TYPE 1	TYPE 2
Hand Crew	Crew #	*Personnel, Equipment, and Transportation	<ul style="list-style-type: none"> Highest training level No use restriction Fully mobilized Highest experience level Fully equipped Permanently assigned supervision 	<ul style="list-style-type: none"> Minimum training or Some use restriction or Not fully mobilized or Moderate experience or Minimum equipment or No assigned supervision
* Indicates <u>minimum</u> number of crew personnel including supervision.			State CDC (12) CYA (12) CCC (12) Fly Crew (8)	Federal (Blue Card) (18) State (12)
			Federal Hotshot (18) Regular (18) Fly Crew (10)	
			Local Govt. Inmate (12) Paid (10)	
			Fly Crew (8) Hotshot (18)	

RESOURCE	RADIO CALL	COMPONENTS	TYPES			
			1	2	3	4
Fire Boat	Boat	Pumping Capability	5,000 GPM	1000 GPM	250 GPM	
Foam Tender	Foam	Class B Foam Specify: % Concentrate (1%, 3%, etc.)	500 Gal.	250 Gal		
Air Tanker	Tanker	Gallons Examples:	3,000 C-130 P-3, DC-7	1,800 DC-4 SP2H,P2V	600 S-2	100 Thrush
Helicopters	Copter	Seats, including pilot Card weight capacity (lbs) Gallon Examples:	16 5000 700 Bell 214	10 2500 300 Bell 204, 205, 212	5 1200 100 Bell 206	3 600 75 Bell 47
Helitanker	Helitanker	- Fixed Tank - Air tanker Board Certified- 1,100 Minimum Gallon Capacity				
Helicopter Tender	Helitender	Fuel Equipment				
Helitack Crew	Helitack	Personnel (3) Equipment Transportation				
Aircraft Rescue Firefighting (ARFF)	ARFF	Class B Foam w/proportioner and pump				



Aircraft name: S-64 Skycrane

Payload: 2,650 gallons

Cruising speed: 91 knots

**Purpose: Used to deliver water or
retardant to slow
progress of fire.**

The S-64 Airplane Helicopter is a powerful and efficient aerial platform. The S-64 is the only purpose-built flying crane in existence. Its unique design features an aft-facing pilot station, which gives the pilot full control of the helicopter and an unobstructed view while placing loads to within inch / millimeter tolerances. Components such as the patented anti-rotational rigging system, hydraulic grapple, high volume fixed fire suppression tank, and 30-second snorkel refill system enhance the Airplane's ability to perform effectively and efficiently, with an external load capacity of 25,000 lbs (11,340 Kg).

S-64E DESIGN, WEIGHTS & PERFORMANCE

GENERAL

Maximum Gross Weight	42,000 lbs (19,051 kg)
Maximum Hook Weight	20,000 lbs (9,072 kg)
Maximum Cruise Speed	115 kts (212 km / hr)
Maximum Range	250 mile (453 km)

FUSELAGE (fuselage is not pressurized)

Length*	69ft 8 in (21.20 m)
Width*	7ft 1 in (2.13 m)
Width Landing Gear	21ft 9.7 in (6.65 m)

* Fuselage only, External Dimension

MAIN ROTOR

Main Rotor Diameter	72ft 2.85 in (22.7 m)
Disc Area	4072 sq. ft (366.5 sq. m)

TAIL ROTOR

Tail Rotor Diameter	16ft (4.90 m)
Disc Area	201.1 sq. ft (18.7 sq. m)

LANDING GEAR

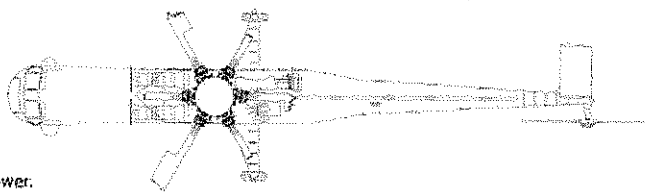
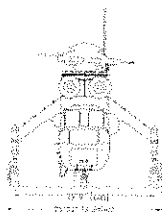
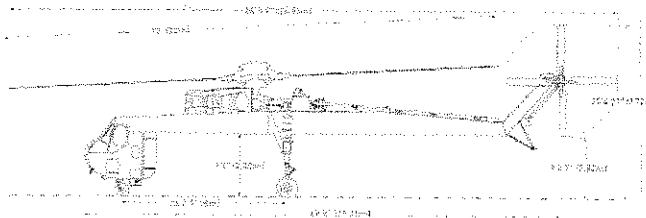
Main Gear	Single Tire
Nose Gear	Single Tire
Wheel Base	24ft 2.7 in (8 m)

POWER PLANT (2 pratt and whitney jftd12a-4a)

Maximum Power Available	9,000 SHP
Single Engine Limit	4,500 SHP / 30 min.

Design Power

For two engine operation at takeoff, the transmission total power is limited to 6,600 SHP for 30 minutes, and must not exceed 5,400 SHP at maximum continuous power.



S-64F DESIGN, WEIGHTS & PERFORMANCE

GENERAL

Maximum Gross Weight	47,000 lbs (21,320 kg)
Maximum Hook Weight	25,000 lbs (11,340 kg)
Maximum Cruise Speed	104 kts (192 km / hr)
Maximum Range	240 mile (444 km)

FUSELAGE (fuselage is not pressurized)

Length*	69ft 8 in (21.20 m)
Width*	7ft 1 in (2.13 m)
Width Landing Gear	21ft 10.7 in (6.67 m)

* Fuselage only, External Dimension

MAIN ROTOR

Main Rotor Diameter	72ft 2.85 in (22.7 m)
Disc Area	4098.2 sq. ft (368.8 sq. m)

TAIL ROTOR

Tail Rotor Diameter	16ft (4.90 m)
Disc Area	201.1 sq. ft (18.7 sq. m)

LANDING GEAR

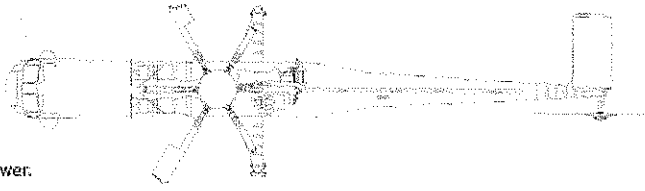
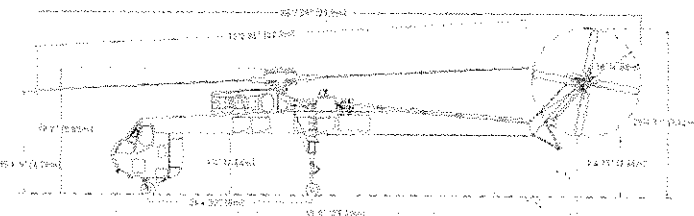
Main Gear	Twin Tires
Nose Gear	Single Tire
Wheel Base	24ft 4.32 in (7.43 m)

POWER PLANT (2 pratt and whitney jftd12a-4a)

Maximum Power Available	9,600 SHP
Single Engine Limit	4,800 SHP / 30 min.

Design Power

For two engine operation at takeoff, the transmission total power is limited to 7,900 SHP for 30 minutes, and must not exceed 7,600 SHP at maximum continuous power.







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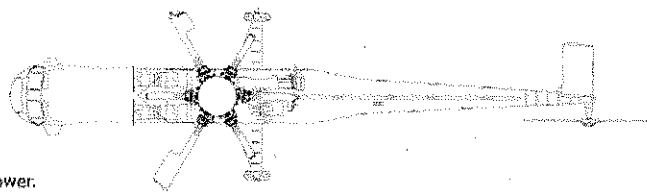
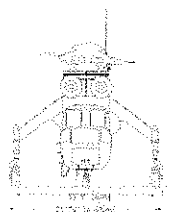
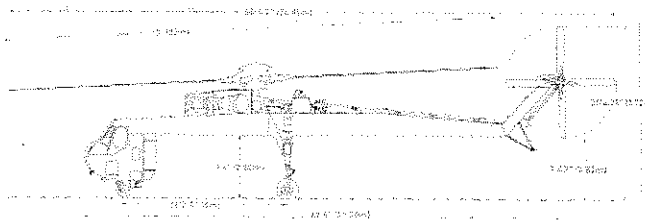
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